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s project is conducted in support of DOE-FE Advanced Combustion and Turbine Programs and DOE-EERE Sunshot Programs. The project is executed bugh NETL Research and Innovation Center's Advanced Combustion Field Work Proposal and CRADA-0588.

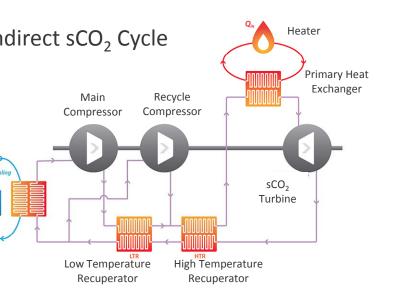
#### IMER

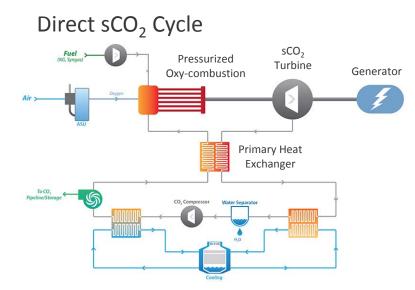
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### upercritical CO<sub>2</sub> Power Cycles







Cycle/Component		Inle	t	Outlet			
		T (C)	P (MPa)	T (C)	P (MPa)		
Indirect	Heater	450-535	1-10	650-750	1-10		
	Turbine	650-750	20-30	550-650	8-10		
	НХ	550-650	8-10	100-200	8-10		
Direct	Combustor	/50	20-30	1150	20-30		
	Turbine	1150	20-30	800	3-8		
	HX	HX 800		100	3-8		

Essentially pure CO<sub>2</sub>

CO<sub>2</sub> with combustion products including O<sub>2</sub>, H<sub>2</sub>O and SO<sub>2</sub>



#### ompact Heat Exchangers



#### Higher efficiency

• Due to much shorter heat diffusion lengths in fluid

#### maller size

- Use of less materials (expensive superalloys)
- Takes less space

#### Modular design

• Expandable to large power plants



# pical Compact HX Fabrication ocess



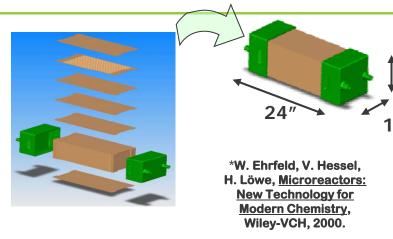
### Pattern microscale flow paths into laminae using a variety of methods

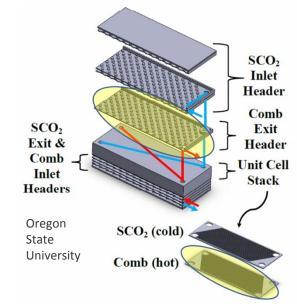
- Chemical etching
- Micromachining
- Laser cutting
- EDM

#### Bond these laminae using a variety of methods

- Diffusion bonding
- Transient liquid phase (TLP) bonding
- Laser welding
- Brazing

sCO<sub>2</sub> cycles, diffusion bonding and TLP bonding considered to be the most robust approaches







### onding



Sonding is considered the "weak link" in the abrication process

harp edges in the architecture lead to ocations of high stress concentration in the nechanical design simulations

#### We need information on

- The parameters for bonding process (T, P, t) of materials
- The strength of the bond
- Corrosion behavior of bonded regions in sCO<sub>2</sub>





#### aterials

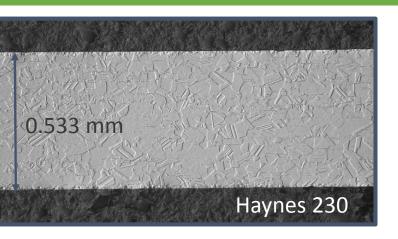
#### High-temperature strength High-temperature oxidation resistance



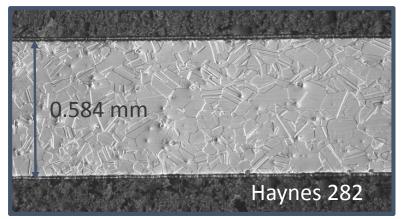
### Nominal chemical composition (weight %) of materials used in this study (Haynes 230 and Haynes 282)

Ni	Cr	W	Ti	Мо	Fe	Со	Mn	Si	Al	С	В
57	22	14		2	3*	5*	0.5	0.4	0.3	0.10	0.015*
57	19.5		2.1	8.5	1.5*	10	0.3*	0.15*	1.5	0.06	0.005

\* = maximum



Solid-solution strengthened Id rolled and 1232 °C solution annealed sheet



Precipitation strengthened 1149 °C solution annealed sheet

## Other materials consider for this application

Inconel 740H
Inconel 625
Inconel 617
347H Stainless stee
316 Stainless stee
304 Stainless stee
Grade 91 steel



### iffusion Bonding vs. Transient Liquid hase (TLP) Bonding



#### **Diffusion Bonding**

Diffusion bonding is a solid tate process

t requires applied high ressure at high temperature or a certain amount of time

t involves diffusion of onstituent atoms and creep rocesses to close the voids resent due to roughness of he faying surfaces.

#### **Transient Liquid Phase Bonding**

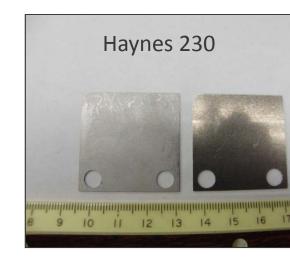
- TLP involves both solid state and liquid state reactions
- It requires less pressure than diffusion bonding
- It requires a lower melting point interlayer
- It involves isothermal melting and solidification of interlayer



### onding



- heets were water-jet cut into shims
- 00 shims were bonded together in each stack
- Ill shims were reverse current etched and cleaned with acetone
- ome stacks used shims plated with electroless ickel, 2 4 µm thick
- ome shims contained pin-fin micro-features dentical to those used in a heat exchanger
- Il shims were thoroughly cleaned by hand and in an ltrasonic acetone bath for 15 minutes immediately efore bonding







### onding



him stacks were held in a fixture during bonding nd pressure was applied only after the emperature ramped up to the desired value

The hot press vacuum was maintained at pproximately 5 x 10<sup>-6</sup> torr (0.0007 Pa)

150°C for 8 hours at 12.7 MPa

Ifter bonding, each stack was machined to roduce 6 tensile specimens using wire EDM and CNC lathe

H282 without Ni plating did not bond well



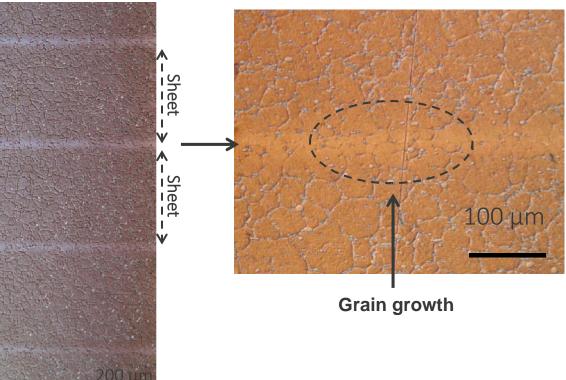




## iffusion Bonding of Alloy 230



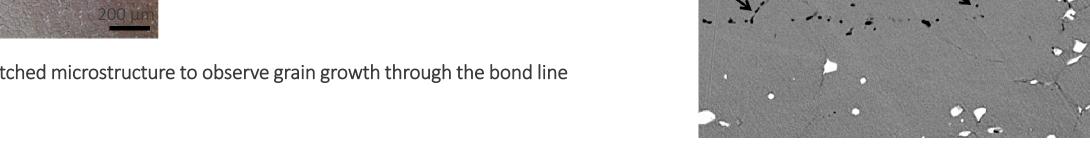
#### Microstructure



Diffusion Bond 50 um

Primary Carbides

Primary carbides form at higher temperature

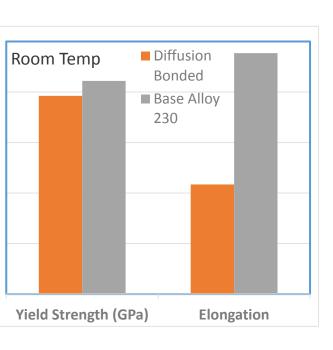


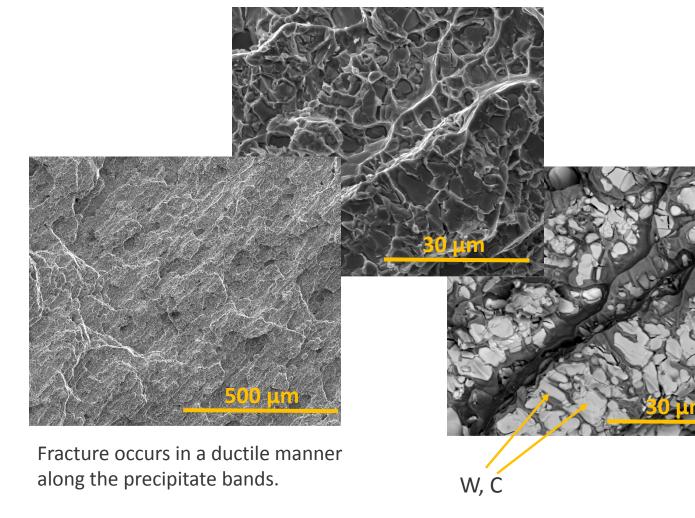


### iffusion Bonding of Alloy 230

#### **Mechanical Behavior**



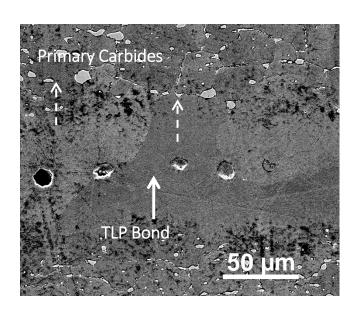






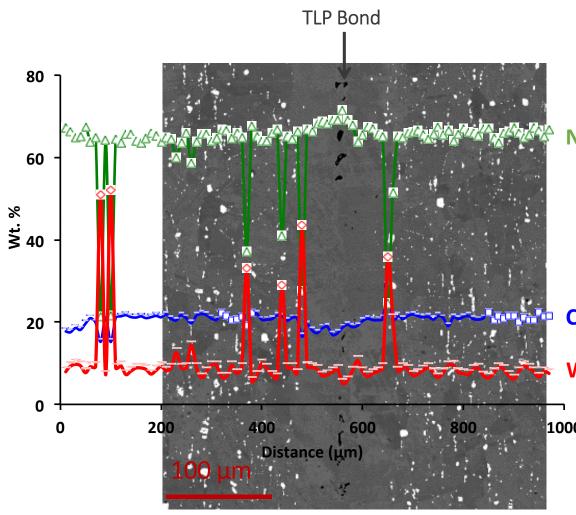
### P Bonding of Alloy 230

#### Microstructure



- Primary Carbides
- Increase in Ni, dip in Cr at the bond

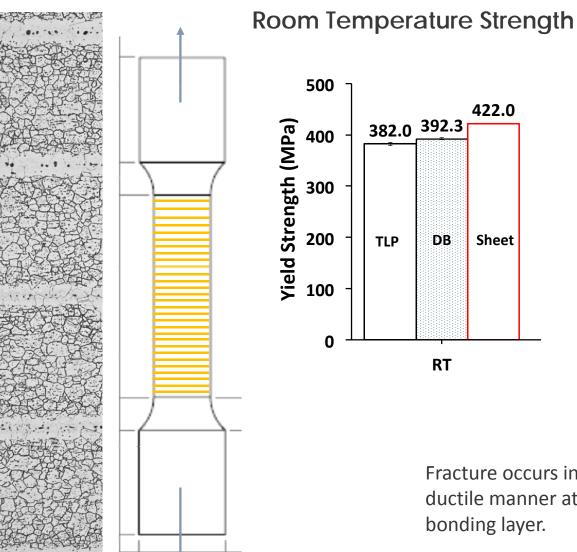


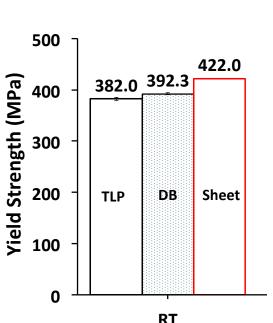




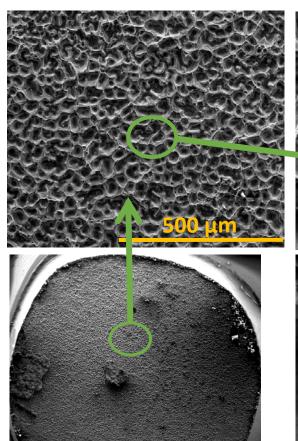
# ond Strength – TLP Bonding of Alloy 230

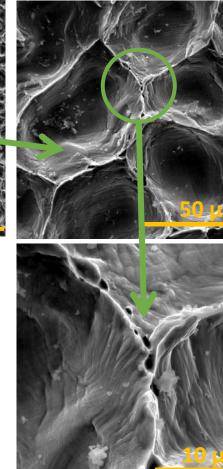






Fracture occurs in a ductile manner at the bonding layer.

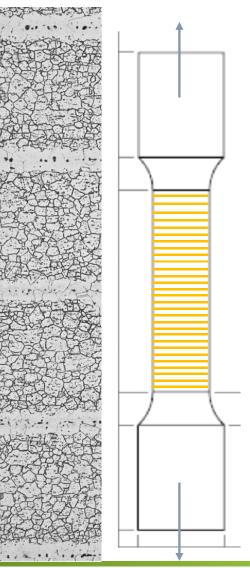




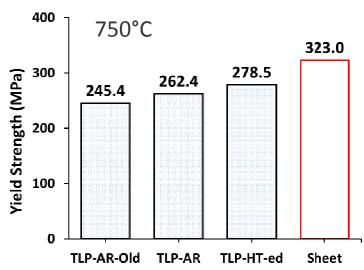


### ond Strength – TLP Bonding of Alloy 230



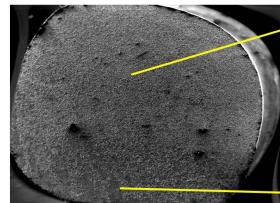


#### **High-Temperature Strength**



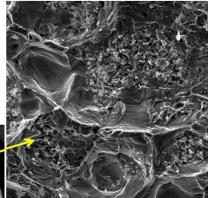
**High Temperature Yield Strength** 

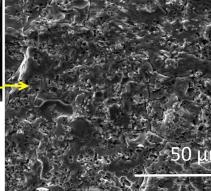
- TLP-AR-Old = 76%
  - TLP-AR = 81%
  - **TLP-HT-ed = 86%**



3 mm

Fracture occurs in a ductile manner at the bonding layer and the sheet.







### onding Defects



#### Challenges with TLP bonding of H282 - Surface oxides & Intermetallic formation

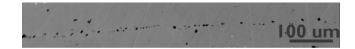
Ni-12P coating on the metal for TLP bonding

**Surface O**xides



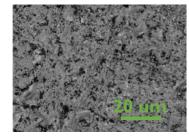
Coating does not adhere to the metal

**Bondline after TLP bonding** 



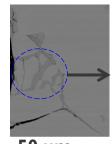
AIO particles (dark) along the bondline

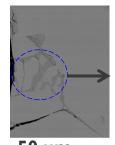
**Fracture Surface** 

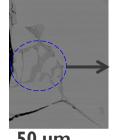


AIO particles (dark) on the fracture surface

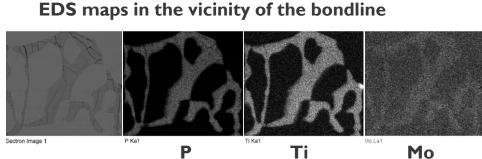
Intermetallic formation due to 12 wt. % P







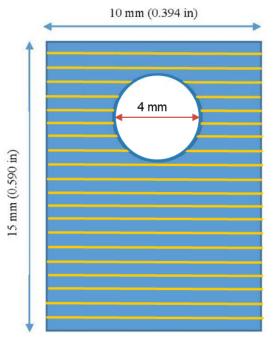


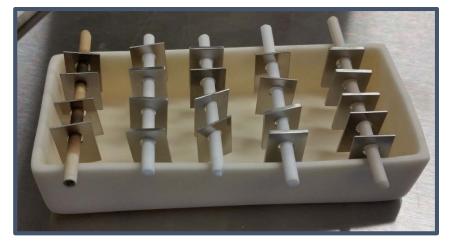


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### xidation of Bonded Regions







Characterizatio
Mass Change
XRD
SEM



Gas flow rate: 0.032 kg/h

Temperature: 700°C

Duration: 2500 h

24 h purging with CO<sub>2</sub> before

heating

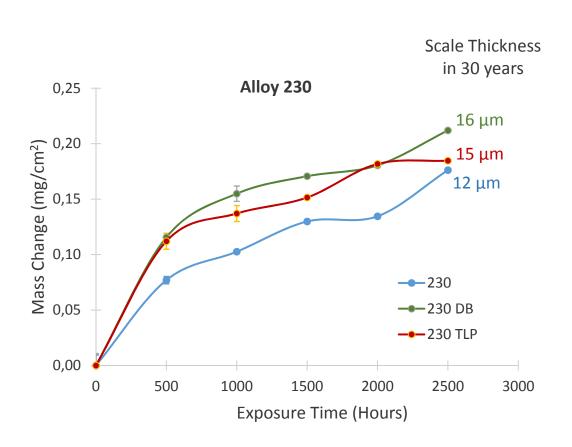


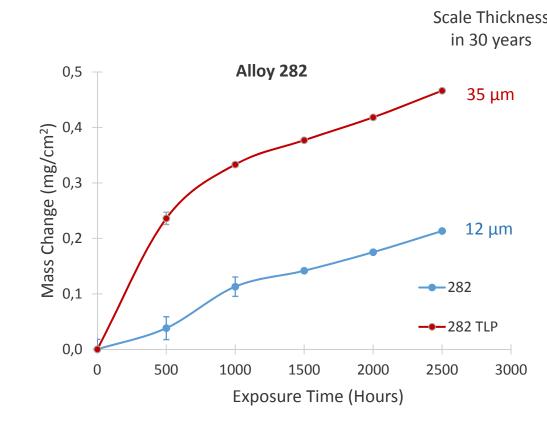


#### xidation of Bonded Regions

bar CO<sub>2</sub> at 700°C





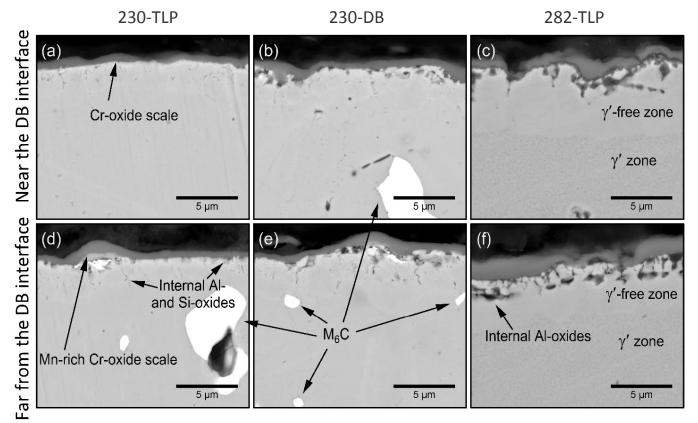




#### xidation of Bonded Regions

bar CO<sub>2</sub> at 700°C





No significant difference between bond regions and away from bond regions

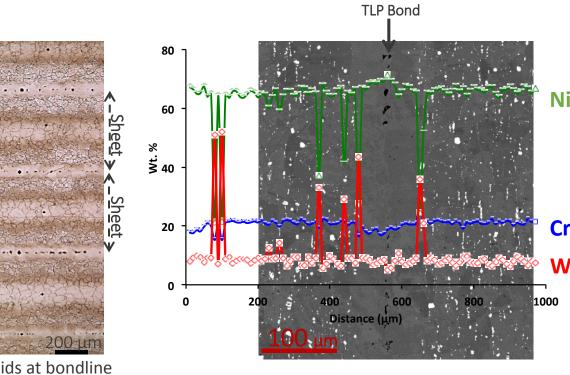
More internal oxidation in H282, resulting from higher Al and Ti levels

γ' loss in H282 below the internal oxidation layer

Back-Scattered Electron Images



#### ımmary



Composition of TLP bonded regions were

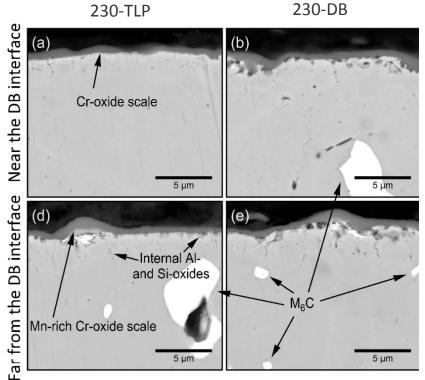
similar to the base material

Ni Cr

400 750°C 323.0 Yield Strength (MPa) 278.5 262.4 245.4 TLP-AR-Old TLP-AR TLP-HT-ed Sheet



Both DB and TLP bonded stacks exhibited good strength at high and room temperature



No signific difference oxidation behavior i CO<sub>2</sub> at 700 between k regions ar away from bond region

